The SOLAR-C Mission: a satellite mission for a high-throughput EUV Imaging Spectroscopy of the Sun



- T. Shimizu, M. Uchiyama, H. Kato, Y. Suematsu, S. Toriumi, H. Bingo, K. Matsuzaki, D. Yamazaki, Y. Kimoto, E. Miyazaki, R. Yamanaka (JAXA)
- H. Hara, R. Ishikawa, F. Uraguchi, T. Oba, T. Okamoto, Y. Katsukawa, Y. Kawabata, M. Kubo, N. Kohara, K. Shinoda, T. Tsuzuki, A. Tei N. Narukage, M. Mitsutake (NAOJ)
- S. Imada (Univ. Tokyo), K. Watanabe (NDA), A. Asai, S. Nagata, T. Yokoyama (Kyoto Univ.), K. Kusano, S. Masuda (Nagoya Univ.) & International SOLAR-C Team

Mission Objectives:

- The SOLAR-C project aims to explore key propositions in space science of how the plasma universe is created and evolves, and
 - how the Sun influences the Earth, other interplanetary objects, and the heliosphere.

Primary Scientific Goals:

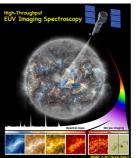
- 1. To understand how fundamental processes lead to the formation of the solar hightemperature atmosphere and the solar wind, and
- 2. To understand how the solar atmosphere becomes unstable, releasing the energy that drives solar flares and eruptions.

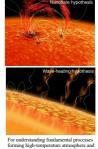
Science Payload:

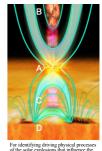
The SOLAR-C observatory has three science payloads:

- EUV Imaging Spectrograph
- >High-throughput performance over each spectral band in EUV
- · UV Slit-jaw imager
- >To watch the spectrograph slit position and monitor the photosphere and chromospheric dynamics
- EUV Solar Spectral Irradiance Monitor (SoSpIM)
- >To monitor the solar EUV spectral irradiance by the Sun as a star approach

These are developed in an international collaboration of JAXA, NASA, ESA, and a few European countries (France, Germany, Italy, and Switzerland). NAOJ (solar physics group + ATC) has extensively contributed to developing the imaging spectrograph.







VNA	solar wind acceleratio	n Earth.	
veł	nicle and satellite	characteristics	
	JAXA Epsilon-S Rocket		

Launch vehicle and satellite characteristics				
Launch Vehicle	aunch Vehicle JAXA Epsilon-S Rocket			
Satellite mass	< 600 kg (nominal)			
Satellite orbit	Sun-synchronous orbit of > 600 km altitude			
Attitude control	three-axis stabilized			

Science Payload Characteristics					
	Imaging Spectrograph	Slit-jaw Imager	SoSpIM(Solar Spectral Irradiance Monitor)		
Telescope:	Primary mirror diameter: 28 cm		No focusing optics:		
offset-parabola primary	Primary mirror fo	cal length: 280 cm	Aperture ϕ 7.6 mm x 3 for each band		
Spatial Resolution (goal)	300 km or 0.4 arcsec	300 km or 0.4 arcsec	Full Sun		
Temporal Resolution	0.5 sec	1 sec	Cadence: 20 Hz		
Observing Wavelengths	17-22 nm 46-128 nm	Continuum: 283.3 nm Mg I: 285.2 nm Mg II: 279.6 nm	Band A (EUV): 17 – 22 nm Band B (Lyα): 112–128 nm		
Field of view	280×280 arcsec ²	280×280 arcsec ²	±0.77 deg		
Wavelength Resolution	λ/Δλ: 5,000-10,000	Band pass $\varDelta\lambda$: ~0.2 nm	Δλ~ 5 (16) nm for Band A (B)		
Plasma T_e to be observed	0.02-15 MK	6,000-10,000 K	Band A (B): 106-7.2 K (104 K)		

